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USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK. VOLUME 148. T-378 IN--ETC(U)

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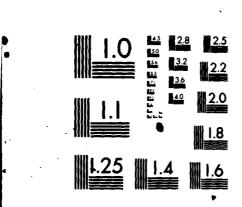
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psychoacoustic measures: overall and band sound pressure levels, C-weighted

and A-weighted sound levels, preferred speech interference level, perceived noise level, and limiting times for total daily exposure of personnel with and

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PREFACE

This report was prepared by the Biodynamic Environment Branch, Air Force Aerospace Medical Research Laboratory, under Project/Task 723108, Crew Safety In Operational Noise Environments.

The author acknowledges the efforts of Mr. John N. Cole who established the data analysis requirements, Mr. Henry Mohlman and Mr. Fred Lampley of the University of Dayton who assisted in the mechanics of data processing and Mrs. Norma Peachey who typed this report and prepared it for publication.

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INTRODUCTION

The T-37B is a USAF two-seat primary trainer aircraft manufactured by the Cessna Aircraft Company, Wichita, Kansas. Power is provided by two Continental J-69-T-25 turbojet engines each rated at 1025 lbs. maximum takeoff thrust. The engines are manufactured by Teledyne CAE, Toledo, Ohio.

This volume provides measured data defining the bioacoustic environments produced inside the aircraft. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with operations of the T-37B aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type, noise data in the handbook describe the noise produced during ground operations of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentations, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Refer to Volume 1 (reference 1) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published, and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of the updated index as it is generated.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

1. Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

IN-FLIGHT NOISE

Measurements

All noise measurements were made on-board a T-37B aircraft during typical speed, altitude, and flight maneuver conditions. These levels describe the standard T-37B environments but may not be representative of those levels encountered if the aircraft has been configured differently (e.g., major equipment or structural changes).

Acoustic measurements were made inside the cockpit at the pilot's location. Table 1 lists the measurement location and test conditions as numeric/alphabetic designators which are used on the data pages. The designator 1/A means measurement location 1 and test condition A, etc.

The microphone was attached to the pilot's helmet by means of a lightweight boom. This arrangement enabled adjustment of the microphone close to the ear level at a distance of 0.1 meter with its diaphragm parallel and facing away from the helmet's surface. In the analysis, microphone corrections for random incidence were applied to the overall systems response. The recorded samples were analyzed using a four or eight second integration time to obtain a power-averaged level which effectively smooths out short duration fluctuations and best describes the exposure.

Results

The measured data presented in Table 2 define the sound pressure levels (SPL) produced inside the T-37B aircraft at the specified location. This table includes the overall, 1/3 octave band, and octave band levels. From these data, C-weighted and A-weighted sound levels, maximum permissible time for one exposure per day (AFR 161-35) with and without standard Air Force ear protectors, preferred speech interference level, and perceived noise level are calculated and presented in Table 3. These measures are widely used to assess the effects of noise on personnel and their performance.

TABLE 1 MEASUREMENT LOCATION AND TEST CONDITIONS T-37B, WRIGHT-PATTERSON AFB, OH; 6 AUG 81 SER NO. 00141

Location	Position	Height Above Deck
1	Co-Pilot	Seated Head Level

Condition	Description
A	Engines Idle, Canopy Closed ECS ON, 39% RPM
В	Engines Idle, Canopy Closed, ECS OFF, 39% RPM
C	Engines Idle, Canopy Open, 39% RPM
Ď	Engines Military, Canopy Closed, ECS ON, 80% RPM
Ē	Engines Military, Canopy Closed, ECS OFF, 80% RPM
$\overline{\mathbf{F}}$	Taxi
Ğ	Takeoff
H	Climb to 6,000 ft
Ī	Cruise 6,000 ft, 220 KIAS, 95% RPM
J	Climb to 20,000 ft, 160 KIAS, 99% RPM
K	Cruise at 20,000 ft, 205 KIAS, 98% RPM, ECS ON
Ï.	Cruise at 20,000 ft, 205 KIAS, 99% RPM, ECS OFF
M	Start of Descent to 10,000 ft
N	Descend to 10,000 ft, at 17,000 ft, speed brakes out
ö	Descend to 10,000 ft, at 13,000 ft, speed brakes up
Ď	Cruise at 10,000 ft, 190 KIAS, 85% RPM, ECS ON
Q	Cruise at 10,000 ft, 190 KIAS, 85% RPM, ECS OFF
Ř	Descend to 6,000 ft
S	Landing Roll

2) IDENTIFICATIONS)) OMEGA 3.2
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6) OMEGA 3.2
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